

Assignment: 2

1. Relevant data pertaining to a helicopter are given in the following.

Weight of the helicopter: 36000 N

Density of air: ρ	1.225 Kg/m ³
Number of blades: N	4
Blade radius: R	6 m
Blade Chord: C	0.4 m
Profile drag coefficient: C_{d0}	0.01
Lift curve slope: a	5.73
Rotor angular rate: Ω	10π rad/sec
Tip loss factor: B	0.97
Root cut-out:	0.15 R

Blade twist for 4 different configurations: $\theta_{tw} = 0$ deg,

-10 deg. (linear twist)

-20 deg. (linear twist)

Ideal twist with θ_{tip}

The helicopter is under hovering condition.

Assuming non-uniform inflow, evaluate the following and show each item in one figure:

- i) Variation of pitch angle with non-dimensional radial location (all 4 twist cases in one figure)
- ii) Variation of angle of attack with non-dimensional radial location (all 4 twist cases in one figure)
- iii) Variation of induced velocity with non-dimensional radial location (all 4 twist cases in one figure)
- iv) Evaluate the Figure of Merit for each twist case. Assume $\kappa=1.0$

Note: Show the plots for non-dimensional radius from 0.15 to 1.0

Solution:









